

Instructor: Alessandra Buonanno (alessandra.buonanno@hu-berlin.de)

Tutor (corresponding for this sheet): Aldo Gamboa (aldo.gamboa@aei.mpg.de)

Tutor: Marcus Haberland (marcus.haberland@aei.mpg.de)

1. On the effective-one-body Hamiltonian and dynamics

We start from Eq (8) of the assignment, relating the real and effective Hamiltonians

$$1 + \hat{H}(q, p)\varepsilon^2 \left(1 + \alpha_1\varepsilon^2\hat{H}(q, p)\right) = \hat{H}_{\text{eff}}(Q, P)\varepsilon^2 \quad (1)$$

where for simplicity we call $\hat{H}_{\text{real}} = \hat{H}$, and we remind that

$$\hat{H}(q, p) = \hat{H}_{\text{Newt}}(q, p) + \varepsilon^2\hat{H}_{\text{1PN}}(q, p) + \dots \quad (2)$$

$$\hat{H}_{\text{Newt}}(q, p) = \frac{1}{2}p^2 - \frac{1}{q}, \quad (3)$$

$$\hat{H}_{\text{1PN}}(q, p) = -\frac{1}{8}(1 - 3\nu)p^4 - \frac{1}{2q}[(3 + \nu)p^2 + \nu(\mathbf{n} \cdot \mathbf{p})^2] + \frac{1}{2q^2}. \quad (4)$$

The canonical transformation at 1PN order has generating function

$$G = q^i p_i \left[c_1 p^i p_i + \frac{c_2}{\sqrt{q^i q_i}} \right] \quad (5)$$

whose derivatives are given by

$$\partial_{p_i} G = q_i \left[c_1 p^2 + \frac{c_2}{q} \right] + q^j p_j (c_1 2p_i) \quad (6)$$

$$\partial_{q_i} G = p^i \left[c_1 p^2 + \frac{c_2}{q} \right] - q^j p_j \left(c_2 \frac{1}{q^3} q^i \right). \quad (7)$$

From the 1PN canonical transformation

$$Q^i = q^i + \frac{1}{c^2} \frac{\partial G_{\text{1PN}}}{\partial p_i}, \quad (8)$$

$$P_i = p_i - \frac{1}{c^2} \frac{\partial G_{\text{1PN}}}{\partial q^i} \quad (9)$$

we can find $Q, P, N_i P_i$ at 1PN (note that $N_i P_i$ will not actually be needed).

$$\begin{aligned} Q &= \sqrt{Q^i Q_i} = \sqrt{q^i q_i + 2q^i \partial_{p_i} G \varepsilon^2} \\ &= \sqrt{q^i q_i + 2\varepsilon^2 \left[q^2 \left(c_1 p^2 + \frac{c_2}{q} \right) + 2(\mathbf{q} \cdot \mathbf{p})^2 c_1 \right]} \end{aligned} \quad (10)$$

$$\begin{aligned} P &= \sqrt{p^i p_i} = \sqrt{p^i p_i - 2p^i \partial_{q_i} G \varepsilon^2} \\ &= \sqrt{p^i p_i - 2\varepsilon^2 \left[p^2 \left(c_1 p^2 + \frac{c_2}{q} \right) - (\mathbf{q} \cdot \mathbf{p})^2 \frac{c_2}{q^3} \right]} \end{aligned} \quad (11)$$

$$\begin{aligned}
N \cdot P &= \frac{Q^i}{Q} P_i = \frac{1}{Q} (\mathbf{p} \cdot \mathbf{q} + \varepsilon^2 (p^i \partial_{p_i} G - q^i \partial_{q_i} G)) \\
&= \frac{1}{Q} \left(\mathbf{p} \cdot \mathbf{q} + \varepsilon^2 \left(\mathbf{p} \cdot \mathbf{q} \left(c_1 p^2 + \frac{c_2}{q} + 2c_1 p^2 \right) - \mathbf{p} \cdot \mathbf{q} \left(c_1 p^2 + \frac{c_2}{q} - \frac{c_2}{q} \right) \right) \right) \\
&= \frac{1}{Q} \left(\mathbf{p} \cdot \mathbf{q} + \varepsilon^2 \left[\mathbf{p} \cdot \mathbf{q} \left(2c_1 p^2 + \frac{c_2}{q} \right) \right] \right) \quad (\text{expand } Q) \\
&\simeq \frac{1}{q} \left(\mathbf{p} \cdot \mathbf{q} + \varepsilon^2 \left[\mathbf{p} \cdot \mathbf{q} \left(2c_1 p^2 + \frac{c_2}{q} \right) \right] \right) \left(1 + 2\varepsilon^2 \left[c_1 p^2 + \frac{c_2}{q} + 2 \left(\frac{\mathbf{q} \cdot \mathbf{p}}{q} \right)^2 c_1 \right] \right)
\end{aligned} \tag{12}$$

Let's match the square of the Hamiltonian order by order:

$$\left[1 + \varepsilon^2 \hat{H} \left(1 + \alpha_1 \varepsilon^2 \hat{H} \right) \right]^2 = \hat{H}_{\text{eff}}^2 \varepsilon^4$$

$$1 + 2\varepsilon^2 \hat{H} \left(1 + \alpha_1 \varepsilon^2 \hat{H} \right) + \varepsilon^4 \hat{H}^2 \left(1 + \alpha_1 \varepsilon^2 \hat{H} \right)^2 = \hat{H}_{\text{eff}}^2 \varepsilon^4 \tag{13}$$

Let's consider first the right hand side:

$$\begin{aligned}
\varepsilon^4 \hat{H}_{\text{eff}}^2 &= \left[\left(1 + \frac{a_1}{Q} \varepsilon^2 \right) \left(1 + \varepsilon^2 P^2 + \varepsilon^4 a_1 \frac{(\mathbf{N} \cdot \mathbf{P})^2}{Q} \right) \right] \\
&= 1 + \varepsilon^2 P^2 + \varepsilon^4 a_1 \frac{(\mathbf{N} \cdot \mathbf{P})^2}{Q} + \frac{a_1}{Q} \varepsilon^2 + \frac{a_1 P^2}{Q} \varepsilon^4 + \mathcal{O}(\varepsilon^6) \\
&= 1 + \varepsilon^2 \left(P^2 + \frac{a_1}{Q} \right) + \varepsilon^4 \left(a_1 \frac{(\mathbf{N} \cdot \mathbf{P})^2}{Q} + \frac{a_1 P^2}{Q} \right) + \mathcal{O}(\varepsilon^6)
\end{aligned} \tag{14}$$

Left hand side:

$$\begin{aligned}
&1 + 2\varepsilon^2 \hat{H} \left(1 + \alpha_1 \varepsilon^2 \hat{H} \right) + \varepsilon^4 \hat{H}^2 \left(1 + \alpha_1 \varepsilon^2 \hat{H} \right)^2 \\
&1 + 2\varepsilon^2 \hat{H} + 2\alpha_1 \hat{H}^2 \varepsilon^4 + \varepsilon^4 \hat{H}^2 + \mathcal{O}(\varepsilon^6) \\
&1 + 2\varepsilon^2 \hat{H} + \varepsilon^4 \hat{H}^2 (2\alpha_1 + 1) + \mathcal{O}(\varepsilon^6) \quad (\hat{H} = \hat{H}_N + \varepsilon^2 H_{1\text{PN}}) \\
&1 + 2\varepsilon^2 \hat{H}_N + \varepsilon^4 \left(2\hat{H}_{1\text{PN}} + \hat{H}_N^2 (2\alpha_1 + 1) \right) + \mathcal{O}(\varepsilon^6)
\end{aligned} \tag{15}$$

Matching at $\mathcal{O}(\varepsilon^2)$:

$$2\hat{H}_N \stackrel{!}{=} P^2 + \frac{a_1}{Q}$$

$$\begin{aligned}
p^2 - \frac{2}{q} &\stackrel{!}{=} p^i p_i - 2\varepsilon^2 \left[p^2 \left(c_1 p^2 + \frac{c_2}{q} \right) - (\mathbf{q} \cdot \mathbf{p})^2 \frac{c_2}{q^3} \right] + \\
&+ a_1 \left(q^i q_i + 2\varepsilon^2 \left[q^2 \left(c_1 \mathbf{p}^2 + \frac{c_2}{q} \right) + 2c_1 (\mathbf{q} \cdot \mathbf{p})^2 \right] \right)^{-\frac{1}{2}} \\
&= p^2 - 2\varepsilon^2 \left[p^2 \left(c_1 p^2 + \frac{c_2}{q} \right) - (\mathbf{q} \cdot \mathbf{p})^2 \frac{c_2}{q^3} \right] \\
&+ \frac{a_1}{q} \left(1 - \varepsilon^2 \left[\frac{q^2}{q^2} \left(c_1 p^2 + \frac{c_2}{q} \right) + \frac{2c_1 (\mathbf{q} \cdot \mathbf{p})^2}{q^2} \right] \right)
\end{aligned} \tag{16}$$

From which we obtain $a_1 = -2$. The $\mathcal{O}(\varepsilon^2)$ terms will also be relevant at the next order.

Matching at $\mathcal{O}(\varepsilon^4)$, the LHS is given by:

$$2 \left(-\frac{1-3\nu}{8} p^4 - \frac{1}{2q} [(3+\nu)p^2 + \nu(\mathbf{n} \cdot \mathbf{p})^2] + \frac{1}{2q^2} \right) + (2\alpha_1 + 1) \left(\frac{p^4}{4} + \frac{1}{q^2} - \frac{p^2}{q} \right) \quad (17)$$

RHS:

$$\left[a_1 \frac{(\mathbf{N} \cdot \mathbf{P})^2}{Q} + \frac{a_1 P^2}{Q} \right] + \left(\frac{-a_1}{q} \right) \left[\left(c_1 p^2 + \frac{c_2}{q} \right) + \frac{2c_1 (\mathbf{q} \cdot \mathbf{p})^2}{q^2} \right] - 2 \left[p^2 \left(c_1 p^2 + \frac{c_2}{q} \right) - (\mathbf{q} \cdot \mathbf{p})^2 \frac{c_2}{q^3} \right] \quad (18)$$

RHS, expanding and putting terms together:

$$a_1 \left(\frac{(\mathbf{n} \cdot \mathbf{p})^2}{q} + \frac{p^2}{q} - \frac{p^2 c_1}{q} - \frac{c_2}{q^2} - 2c_1 \frac{(\mathbf{q} \cdot \mathbf{p})^2}{q^3} \right) - 2 \left(c_1 p^4 + c_2 \frac{p^2}{q} - \frac{(\mathbf{q} \cdot \mathbf{p})^2}{q^3} c_2 \right) = \quad (19)$$

$$\left[\frac{(\mathbf{n} \cdot \mathbf{p})^2}{q} (-2 + 2c_2 - 2(-2c_1)) \right] + \left[\frac{p^2}{q} (-2 + 2c_1 - 2c_2) \right] + \left[\frac{2c_2}{q^2} - 2c_1 p^4 \right]$$

LHS:

$$p^4 \left[-\frac{1-3\nu}{4} + \frac{2\alpha_1 + 1}{4} \right] + [2\alpha_1 + 1 + 1] \frac{1}{q^2} \quad (20)$$

$$+ [- (2\alpha_1 + 1) - 3 - \nu] \frac{p^2}{q} + [-\nu] \frac{(\mathbf{n} \cdot \mathbf{p})^2}{q}$$

By matching the coefficients of the corresponding terms, and remembering that $\frac{(\mathbf{n} \cdot \mathbf{p})^2}{q} = \frac{(\mathbf{q} \cdot \mathbf{p})^2}{q^3}$, we obtain the following equations:

$$c_2 = \alpha_1 + 1 \quad (21)$$

$$-2c_1 = \frac{2\alpha_1 + 1 - 1 + 3\nu}{4} = \frac{2\alpha_1 + 3\nu}{4} \implies c_1 = -\frac{2\alpha_1 + 3\nu}{8} \quad (22)$$

$$-2 + 2c_2 - 2(-2c_1) = -\nu \rightarrow +4\alpha_1 - 2\alpha_1 - 3\nu = -2\nu \implies \alpha_1 = \frac{\nu}{2} \quad (23)$$

$$c_1 = -\frac{\nu}{2}, \quad c_2 = 1 + \frac{\nu}{2} \quad (24)$$

2. Incorporating the emission of gravitational waves in the two-body dynamics:

- a) For the energy-flux relation, we calculate the total time derivative of the Hamiltonian with the chain rule:

$$\frac{dH}{dt} = \frac{\partial H}{\partial t} + \frac{\partial H}{\partial r} \dot{r} + \frac{\partial H}{\partial \phi} \dot{\phi} + \frac{\partial H}{\partial p_r} \dot{p}_r + \frac{\partial H}{\partial p_\phi} \dot{p}_\phi. \quad (25)$$

If the Hamiltonian does not depend explicitly on time or in the azimuthal angle ϕ , we can simplify this to

$$\frac{dH}{dt} = \frac{\partial H}{\partial r} \dot{r} + \frac{\partial H}{\partial p_r} \dot{p}_r + \frac{\partial H}{\partial p_\phi} \dot{p}_\phi. \quad (26)$$

Next, we substitute Hamilton's EOMs to get:

$$\frac{dH}{dt} = \frac{\partial H}{\partial r} \frac{\partial H}{\partial p_r} + \frac{\partial H}{\partial p_r} \left(-\frac{\partial H}{\partial r} + \mathcal{F}_r \right) + \frac{\partial H}{\partial p_\phi} \mathcal{F}_\phi \quad (27)$$

$$= \dot{r} \mathcal{F}_r + \dot{\phi} \mathcal{F}_\phi, \quad (28)$$

which is the desired result since $dH/dt = -\Phi_E$.

The angular-momentum-flux relation is directly obtained by balancing the change in the angular momentum (given by Hamilton's EOMs) with the angular momentum flux:

$$\dot{p}_\phi = -\Phi_L = \mathcal{F}_\phi. \quad (29)$$

- b) To compute the orbit averages, we would need to obtain some relations within the Keplerian parametrization. First, we note that the orbit average of a function f over an orbital period can be calculated as

$$\langle f \rangle \equiv \frac{1}{T} \oint f dt = \frac{1}{T} \int_0^{2\pi} \frac{f}{\dot{\phi}} d\phi, \quad (30)$$

where $T = \oint dt = \int_0^{2\pi} 1/\dot{\phi} d\phi$. To compute these integrals, we employ the relation between the angular momentum and the angular velocity of the orbit:

$$\dot{\phi} = \frac{L}{\mu r^2}, \quad (31)$$

where

$$r = \frac{R}{1 + e \cos \phi} = \frac{a(1 - e^2)}{1 + e \cos \phi}. \quad (32)$$

We will also need the following relations from the Kepler's problem:

$$R = \frac{L^2}{GM\mu^2}, \quad (33)$$

$$e = 1 + \frac{2EL^2}{GM^2\mu^3}, \quad (34)$$

where E is the energy of the orbit.

Employing these relations, we first get

$$T = 2\pi \sqrt{\frac{a^3}{GM}}. \quad (35)$$

Now, we move on to calculate the quadrupole moment Q_{ij} of the source:

$$Q_{ij} = M_{ij} - \frac{1}{3} \delta_{ij} \text{Tr}(M), \quad (36)$$

$$M_{ij} = \mu x_i x_j = \mu r^2 \begin{pmatrix} \cos^2 \phi & \sin \phi \cos \phi & 0 \\ \sin \phi \cos \phi & \sin^2 \phi & 0 \\ 0 & 0 & 1 \end{pmatrix}, \quad (37)$$

where we have used a polar parametrization of the orbit, such that $x = r \cos \phi$ and $y = r \sin \phi$.

To take the time derivatives of the quadrupole, we can use three times (for the three time derivatives) the relation

$$\dot{Q}_{ij} = \frac{dQ_{ij}}{d\phi} \dot{\phi}, \quad (38)$$

where the r inside Q_{ij} is expressed via Eq. (32), and in this way, we will get expressions as functions of ϕ , which can be integrated to get the orbit-averages.

An alternative (more generic) approach is to take the time derivatives of the multipole moments *without* specifying the Keplerian parametrization of the orbit, and then, every time a time derivative appears ($\dot{r}, \dot{\phi}, \dot{p}_r, \dot{p}_\phi$), we substitute Hamilton's EOMs for the Kepler problem. In the attached `Mathematica` solution, we follow the latter strategy, and we work in reduced variables.

The solution, in physical units, is found to be

$$\langle \Phi_E \rangle = \frac{32 G^4 \mu^2 M^3}{5 c^5 a^5} \frac{1}{(1-e^2)^{7/2}} \left(1 + \frac{73}{24} e^2 + \frac{37}{96} e^4 \right), \quad (39a)$$

$$\langle \Phi_L \rangle = \frac{32 G^{7/2} \mu^2 M^{5/2}}{5 c^5 a^{7/2}} \frac{1}{(1-e^2)^2} \left(1 + \frac{7}{8} e^2 \right). \quad (39b)$$

- c) Since energy is conserved along the orbit, then we evaluate the Hamiltonian at the two turning points $\phi = 0, \pi$ where $p_r = 0$, and then we can solve the following system of equations in terms of p_ϕ :

$$E = \frac{p_\phi^2}{2r_+^2} - \frac{1}{r_+}, \quad (40)$$

$$E = \frac{p_\phi^2}{2r_-^2} - \frac{1}{r_-}, \quad (41)$$

where

$$r_+ = \frac{R}{1+e} \quad \text{and} \quad r_- = \frac{R}{1-e}. \quad (42)$$

Doing this, and employing the relation $R = a(1-e^2)$ we find

$$p_\phi = L = \sqrt{a(1-e^2)}. \quad (43)$$

Evaluating this relation in Eq. (40), we get

$$E = -\frac{1}{2a}. \quad (44)$$

Finally, using Eqs. (43) and (44) in the generic expression of the Hamiltonian, we get

$$p_r = \frac{e \sin \phi}{\sqrt{a(1-e^2)}}. \quad (45)$$

- d) This problem is solved by direct integration of the involved quantities. We require Hamilton's EOM for $\dot{\phi}$ and \dot{r} , as well as the expressions for r , p_ϕ , and p_r given by Eqs. (32), (43), and (45), respectively. The orbit average is computed with the relation given in Eq. (30). Since this is a straightforward process, we do it in the attached `Mathematica` notebook, where it is shown that we recover the expected results.
- e) This problem is also treated in the attached `Mathematica` notebook. Below is shown a plot of the trajectory.

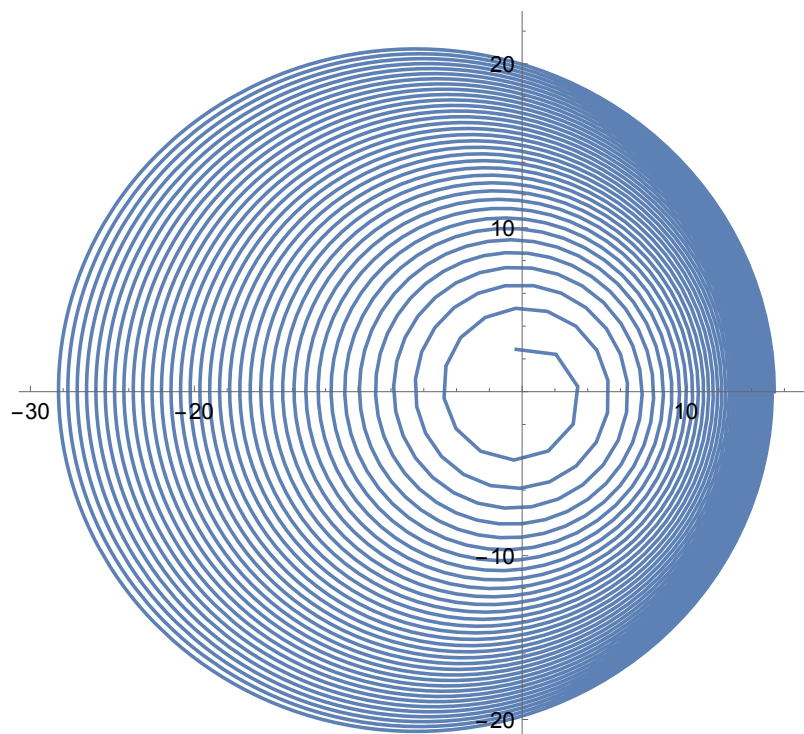


FIG. 1. Trajectory in the xy -plane for an equal-mass binary with starting values of $e = 0.3$, $R = 20$, $\phi = 0$, $\alpha = -16/3$, and $\beta = -13/2$.

Solution to HW4

GW course 2024

Prof. Alessandra Buonanno

TAs: Aldo Gamboa and Marcus Haberland

```
In[1]:= $Assumptions = {a > 0, e > 0, e < 1};
```

2b)

```
In[2]:= addtdependence = {ϕ → ϕ[t], r → r[t], pr → pr[t], pϕ → pϕ[t]};  
removetdependence = {ϕ[t] → ϕ, r[t] → r, pr[t] → pr, pϕ[t] → pϕ};
```

We define the Newtonian Hamiltonian (in reduced variables):

```
In[4]:= H[r_, pr_, pϕ_] := pr^2 / 2 + pϕ^2 / (2 r^2) - 1 / r
```

Then Hamilton's equations are:

```
In[5]:= dotϕ = D[H[r, pr, pϕ], pϕ]
```

```
Out[5]=  $\frac{p\phi}{r^2}$ 
```

```
In[6]:= dotr = D[H[r, pr, pϕ], pr]
```

```
Out[6]= pr
```

```
In[7]:= dotpr = - D[H[r, pr, pϕ], r]
```

```
Out[7]=  $\frac{p\phi^2}{r^3} - \frac{1}{r^2}$ 
```

```
In[8]:= substituteEOMs = {r'[t] → dotr, ϕ'[t] → dotϕ, pr'[t] → dotpr, pϕ'[t] → 0};
```

We compute the quadrupole moment and its time derivatives

```
In[9]:= M = r^2 {{Cos[ϕ]^2, Sin[ϕ] Cos[ϕ], 0}, {Sin[ϕ] Cos[ϕ], Sin[ϕ]^2, 0}, {0, 0, 1}};  
Q = M - 1 / 3 IdentityMatrix[3] × Tr[M] // Simplify
```

```
Out[10]=
```

```
{ {  $\frac{1}{6} r^2 (-1 + 3 \cos[2 \phi])$ ,  $r^2 \cos[\phi] \sin[\phi]$ , 0 },  
  {  $r^2 \cos[\phi] \sin[\phi]$ ,  $-\frac{1}{6} r^2 (1 + 3 \cos[2 \phi])$ , 0 }, { 0, 0,  $\frac{r^2}{3}$  } }
```

In[11]:= **dQdt = D[Q /. addtdependence, t] /. removetdependence /. substituteEOMs // FullSimplify**

Out[11]=

$$\left\{ \left\{ -\frac{pr r}{3} + pr r \cos[2\phi] - p\phi \sin[2\phi], p\phi \cos[2\phi] + pr r \sin[2\phi], 0 \right\}, \right. \\ \left. \left\{ p\phi \cos[2\phi] + pr r \sin[2\phi], -\frac{1}{3} pr r (1 + 3 \cos[2\phi]) + p\phi \sin[2\phi], 0 \right\}, \left\{ 0, 0, \frac{2pr r}{3} \right\} \right\}$$

In[12]:= **d2Qdt2 = D[dQdt /. addtdependence, t] /. removetdependence /. substituteEOMs // FullSimplify**

Out[12]=

$$\left\{ \left\{ -\frac{1}{3r^2} (p\phi^2 + r(-1 + pr^2 r) + 3(p\phi^2 + r - pr^2 r^2) \cos[2\phi] + 6pr p\phi r \sin[2\phi]), \right. \right. \\ \left. \frac{2pr p\phi r \cos[2\phi] - (p\phi^2 + r - pr^2 r^2) \sin[2\phi]}{r^2}, 0 \right\}, \\ \left\{ \frac{2pr p\phi r \cos[2\phi] - (p\phi^2 + r - pr^2 r^2) \sin[2\phi]}{r^2}, \right. \\ \left. \frac{-p\phi^2 + r - pr^2 r^2 + 3(p\phi^2 + r - pr^2 r^2) \cos[2\phi] + 6pr p\phi r \sin[2\phi]}{3r^2}, 0 \right\}, \\ \left\{ 0, 0, \frac{2(p\phi^2 + r(-1 + pr^2 r))}{3r^2} \right\} \right\}$$

In[13]:= **d3Qdt3 = D[d2Qdt2 /. addtdependence, t] /. removetdependence /. substituteEOMs // FullSimplify**

Out[13]=

$$\left\{ \left\{ \frac{pr r - 3pr r \cos[2\phi] + 12p\phi \sin[2\phi]}{3r^3}, -\frac{4p\phi \cos[2\phi] + pr r \sin[2\phi]}{r^3}, 0 \right\}, \right. \\ \left\{ -\frac{4p\phi \cos[2\phi] + pr r \sin[2\phi]}{r^3}, \frac{pr r + 3pr r \cos[2\phi] - 12p\phi \sin[2\phi]}{3r^3}, 0 \right\}, \\ \left\{ 0, 0, -\frac{2pr}{3r^2} \right\} \right\}$$

Now, we compute the instantaneous energy flux:

$1/5 d^3 Q_{ij}/dt^3 d^3 Q_{ij}/dt^3$:

In[14]:= **EE = 1/5 Sum[d3Qdt3[[i, j]] × d3Qdt3[[i, j]], {i, 1, 3}, {j, 1, 3}] // FullSimplify**

Out[14]=

$$\frac{8(12p\phi^2 + pr^2 r^2)}{15r^6}$$

And we compute the instantaneous angular momentum flux:

$2/5 \epsilon^{3ij} d^2 Q_{ij}/dt^3 d^3 Q_{ij}/dt^3$:


```
In[15]:=  $\bar{L} = 2 / 5 (\text{Sum}[d2Qdt2[[1, i]] \times d3Qdt3[[2, i]], \{i, 1, 3\}] -$   

 $\text{Sum}[d2Qdt2[[2, i]] \times d3Qdt3[[1, i]], \{i, 1, 3\}]) // \text{FullSimplify}$ 
```

```
Out[15]=
```

$$\frac{8 p \phi (2 p \phi^2 + r (2 - p r^2 r))}{5 r^5}$$

Now, we move on to compute the orbit-average.

First we need the Keplerian transformations:

```
In[16]:= toKeplerian = {r → a (1 - e^2) / (1 + e Cos[φ]),  

pr → e Sin[φ] / Sqrt[a (1 - e^2)], pφ → Sqrt[a (1 - e^2)]};
```

Next, we obtain the period of the orbit:

```
In[17]:= period = Integrate[1 / dotφ /. toKeplerian // FullSimplify, {φ, 0, 2 Pi}]
```

```
Out[17]=
```

$$2 a^{3/2} \pi$$

In this way, we compute the orbit-averaged energy:

```
In[18]:= 1 / period Integrate[ $\bar{E}$  / dotφ /. toKeplerian // FullSimplify, {φ, 0, 2 Pi}] /.  

Sqrt[a - a e^2] → Sqrt[a] Sqrt[1 - e^2] // FullSimplify
```

```
Out[18]=
```

$$\frac{96 + 292 e^2 + 37 e^4}{15 a^5 (1 - e^2)^{7/2}}$$

And the orbit-averaged angular momentum:

```
In[19]:= 1 / period Integrate[ $\bar{L}$  / dotφ /. toKeplerian // FullSimplify, {φ, 0, 2 Pi}] /.  

Sqrt[a - a e^2] → Sqrt[a] Sqrt[1 - e^2] // FullSimplify
```

```
Out[19]=
```

$$\frac{4 (8 + 7 e^2)}{5 a^{7/2} (-1 + e^2)^2}$$

2d)

We define the Newtonian Hamiltonian (in reduced variables):

```
In[20]:= H[r_, pr_, pφ_] := pr^2 / 2 + pφ^2 / (2 r^2) - 1 / r
```

Then Hamilton's equations are:

```
In[21]:= dotφ = D[H[r, pr, pφ], pφ]
```

```
Out[21]=
```

$$\frac{p \phi}{r^2}$$

```
In[22]:= dotr = D[H[r, pr, pφ], pr]
```

```
Out[22]=
```

$$pr$$

```
In[23]:= dotpr = - D[H[r, pr, pφ], r]
```

```
Out[23]=
```

$$\frac{p\phi^2}{r^3} - \frac{1}{r^2}$$

We define the RR force components:

```
In[24]:= Fr[r_, pr_, pφ_] := 8 √ pr / (15 r^3)
```

```
      ((-3 α + 9 β + 3) (pr^2 + pφ^2 / r^2) + (9 α - 15 β + 9) pr^2 + (9 α - 9 β + 17) / r);
```

```
Fφ[r_, pr_, pφ_] :=
```

```
      8 √ pφ / (15 r^3) (9 (α + 1) pr^2 - 3 (2 + α) (pr^2 + pφ^2 / r^2) + 3 (α - 2) / r);
```

The transformation to Keplerian parametrization is:

```
In[26]:= toKeplerian = {r → a (1 - e^2) / (1 + e Cos[φ]),
```

```
      pr → e Sin[φ] / Sqrt[a (1 - e^2)], pφ → Sqrt[a (1 - e^2)]};
```

And we compute the following quantities in the Keplerian parametrization:

```
In[27]:= integrandE = dotr Fr[r, pr, pφ] + dotφ Fφ[r, pr, pφ] /. toKeplerian // FullSimplify
```

```
Out[27]=
```

$$\frac{1}{15 a^5 (-1 + e^2)^5} \sqrt{(1 + e \cos[\phi])^3} \\ \left(96 - 4 e^2 (-31 + 3 \alpha + 3 e^2 (2 + \alpha)) + 9 e^4 \beta + 2 e (12 (14 + \alpha) + e^2 (61 + 12 \alpha - 9 \beta)) \cos[\phi] + \right. \\ \left. 4 e^2 (77 + 21 \alpha + 3 e^2 (6 + 3 \alpha - 2 \beta)) \cos[2 \phi] + \right. \\ \left. e^3 (2 (59 + 24 \alpha + 9 \beta) \cos[3 \phi] + 15 e \beta \cos[4 \phi]) \right)$$

```
In[28]:= integrandL = Fφ[r, pr, pφ] /. toKeplerian // FullSimplify
```

```
Out[28]=
```

$$-\frac{1}{5 a^4 (-1 + e^2)^4} \\ 4 \sqrt{a - a e^2} \sqrt{(1 + e \cos[\phi])^3} (8 + e^2 - e^2 \alpha + 2 e (6 + \alpha) \cos[\phi] + 3 e^2 (1 + \alpha) \cos[2 \phi])$$

Now, we perform the orbit-averages to obtain the fluxes:

```
In[29]:= period = Integrate[1 / dotφ /. toKeplerian // FullSimplify, {φ, 0, 2 Pi}]
```

```
Out[29]=
```

$$2 a^{3/2} \pi$$

```
In[30]:= orbAvL =
```

```
      - 1 / period Integrate[integrandL / dotφ /. toKeplerian // FullSimplify, \\ \phi, 0, 2 Pi]] // FullSimplify
```

```
Out[30]=
```

$$\frac{4 (8 + 7 e^2) \sqrt{}}{5 a^{7/2} (-1 + e^2)^2}$$

```
In[31]:= orbAvE =
  - 1 / period Integrate[integrandE / dotφ /. toKeplerian // FullSimplify,
    {φ, 0, 2 Pi}] // FullSimplify
```

```
Out[31]=

$$\frac{(96 + 292 e^2 + 37 e^4) \nu}{15 a^5 (1 - e^2)^{7/2}}$$

```

2 e)

With the code below, we can play with the parameters to get interesting trajectories

```
In[32]:= addtdependence = {φ → φ[t], r → r[t], pr → pr[t], pφ → pφ[t]};
numvalues = {α → -16 / 3, β → -13 / 2, ν → 0.25};
initvalues = {e → 0.3, φ → 0, R → 20};
```

```
In[35]:= r0 = R / (1 + e Cos[φ]) /. initvalues
pr0 = e Sin[φ] / Sqrt[R] /. initvalues
pφ0 = Sqrt[R] /. initvalues
```

```
Out[35]=
15.3846
```

```
Out[36]=
0.
```

```
Out[37]=
 $2 \sqrt{5}$ 
```





```

In[38]:= tfinal = 12960; (* This time is selected by exploring larger times,
and realizing that the numerical solver breaks at some point,
so we choose some time before it breaks *)
s = NDSolve[
  {
    r'[t] == (D[H[r, pr, pφ], pr] /. addtdependence /. numvalues),
    φ'[t] == (D[H[r, pr, pφ], pφ] /. addtdependence /. numvalues),
    pr'[t] ==
      (-D[H[r, pr, pφ], r] + Fr[r, pr, pφ] /. addtdependence /. numvalues),
    pφ'[t] == (Fφ[r, pr, pφ] /. addtdependence /. numvalues),
    r[0] == r0,
    φ[0] == 0,
    pr[0] == pr0,
    pφ[0] == pφ0
  },
  {r, φ, pr, pφ},
  {t, tfinal}
]

```

Out[39]=

```

{{r → InterpolatingFunction[ Domain: {{0., 13000.}}, Output: scalar],
 φ → InterpolatingFunction[ Domain: {{0., 13000.}}, Output: scalar],
 pr → InterpolatingFunction[ Domain: {{0., 13000.}}, Output: scalar],
 pφ → InterpolatingFunction[ Domain: {{0., 13000.}}, Output: scalar]}]}

```

```
In[40]:= ParametricPlot[Evaluate[{r[t] Cos[φ[t]], r[t] Sin[φ[t]]} /. s], {t, 0, tfinal}]  
Out[40]=
```

